

Worked Example #7

Calculate the ideal **Chamber Pressure** for a rocket motor with $Kn = 180$. The nozzle exit pressure is 1 atmosphere (optimum expansion). The propellant is KNSU.

The equation for ideal chamber pressure is

$$P_o = \left[\frac{A_b}{A^*} \frac{a \rho_p}{\sqrt{\frac{k}{R T_o} \left(\frac{2}{k+1} \right)^{\frac{k+1}{k-1}}}} \right]^{\frac{1}{(1-n)}} \quad \text{equation 11}$$

From Technical Notepad #3 (<http://www.nakka-rocketry.net/techs.html>), KNSU has the following properties:

$$k = 1.133 \quad \text{for mixture}$$

$$T_o = 1720 \text{ K.}$$

$$M = 41.98 \text{ kg/kmol}$$

The universal gas constant, $R' = 8314 \text{ N-m/kmol-K}$

Noting that $R = R'/M$ giving $R = 198 \text{ N-m/kg-K}$

The propellant density (ρ_p) and burn rate parameters (a, n) are obtained from the KNSU Chemistry web page:

(<http://www.nakka-rocketry.net/succhem.html>)

$$\rho_p = 1.89 \text{ gram/cm}^3$$

Burn rate coefficient, $a = 0.0665 \text{ in}/(\text{sec-psi}^n)$

Burn rate pressure exponent, $n = 0.319$

As the equation for chamber pressure is rather cumbersome, the suggested first step is to simplify the calculation by calculating the terms involving k and n

$$\frac{1}{1-n} = \frac{1}{1-0.319} = 1.468$$

$$\frac{2}{k+1} = \frac{2}{1.133+1} = 0.9376$$

$$\frac{k+1}{k-1} = \frac{1.133+1}{1.133-1} = 16.04$$

Since we wish to express the chamber pressure in SI units, we must use consistent units. We will use *m-k-s* (metre-kilogram-second) units for all parameters:

$$\rho_p = 1.89 \times 1000 = 1890 \text{ kg/m}^3$$

The pressure exponent, n , is dimensionless and can be used as is. The pressure coefficient, a , must be converted to SI units:

$$a = 0.0665 \times \frac{1}{39.37} \times \frac{1}{(6895)^{0.319}} = 0.000101 \frac{m}{sec} \frac{1}{Pa^n}$$

Note 1: $1 \text{ Pa (Pascal)} = 1 \text{ N/m}^2$
 $1 \text{ m} = 39.37 \text{ inches}$
 $1 \text{ psi} = 6895 \text{ Pa}$

Note 2: for more details on conversion method, see
http://www.nakka-rocketry.net/articles/conversion_a.gif)

The ideal chamber pressure may now be calculated:

$$P_o = \left[180 \frac{0.000101(1890)}{\sqrt{\frac{1.133}{198(1720)}(0.9376)^{16.04}}} \right]^{1.468} = 4029044 \text{ Pa} = \underline{\underline{4.03 \text{ MPa}}}$$

To convert to *psi* we divide *Pascals* by 6895 giving $P_o = \underline{\underline{584 \text{ psi}}}$

It is wise to check units for consistency:

$$\left(\frac{m}{sec} \frac{m^{2n}}{N^n} \frac{kg}{m^3} \frac{m}{sec} \right)^{\frac{1}{1-n}}$$

Collecting m and sec terms:

$$\left(\frac{m^{2n-1}}{sec^2} \frac{kg}{N^n} \right)^{\frac{1}{1-n}}$$

Since $kg = N \text{ sec}^2 / m$

$$(m^{2n-2} N^{1-n})^{\frac{1}{1-n}}$$

Simplifying gives units of N/m^2 , which is correct.