

Personal Reflections on Amateur Experimental Rocketry

presented by
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<http://nakka-rocketry.net>

Amateur Experimental Rocketry (AER)

- Rocket motors and propellants are self-designed and self-made.
- Propellant formulations are often unique to AER or “experimental” in nature. New propellant development
- Most or all components of the rocket are self-built

History of AER

- Many of history's great rocket engineers started out as **amateur rocket builders**.

- Robert H. Goddard – launched world's first liquid propellant rocket
- Sergei P. Korolev – A member of GIRD, launched world's first hybrid rocket
- Werhner von Braun – Saturn V moon rocket
- Herman Oberth, Johannes Winkler, Willy Ley & others
- William H. Colburn – invented the “sugar” propellant in 1943.
- Vernon Estes

Robert H. Goddard

1931 Test Rocket # 73



Sergei P. Korolev

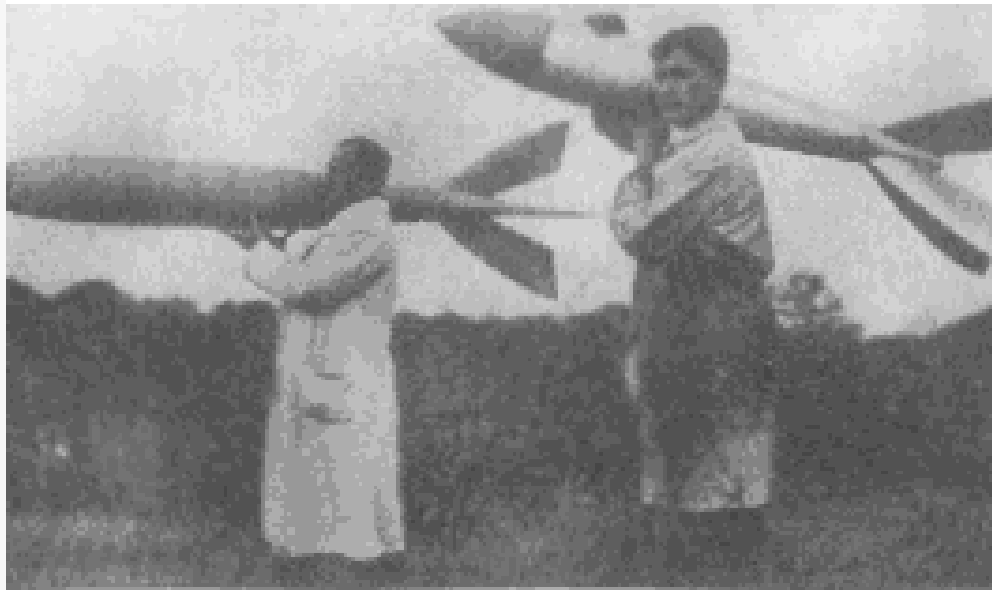
1933 GIRD-X rocket



Members of GIRD with the GIRD-X rocket. Sergei Korolev is at the top left. (Novosti photo, also available as Smithsonian

Wernher von Braun

1930 Mirak rocket



Wernher von Braun in 1930, carrying the Mirak rocket to a test at the Raketenflugplatz in Berlin. At left is Rudolf Nebel. *(Courtesy of the National Air and Space Museum)*

William H. Colburn

1956 Sugar propellant rocket



My early experiences with hobby rocketry

- 1971 - built and flew **model rockets**



UPDATED
SATURN-1
SPECTACULAR
FLYING SCALE
MODEL
formerly SATURN 1B

**4 ENGINE
POWER**

\$9.50

**3 Parachute
Recovery**

A thrilling scale model for even the most advanced rocketeer, this bird will be a real challenge to your building and flying skill. A four engine cluster lifts the Saturn 1-B model into the air. Two 24" parachutes on the main body and a 12" parachute on the command-escape structure return it gently. A real beauty, both in the air and on display. A 12 volt car battery and launch control system (such as the Estes 12 volt Launch Control System) is required for ignition. Kit comes complete with all parts, decals, detailed instructions and a copy of Technical Report No. TR-6 (but no engines). Shipping weight 4 lbs.

Cat. No. 671-K-29...\$9.50

Specifications
Length.....37"
Body Dia.....3.53"
Weight.....9.86 oz.

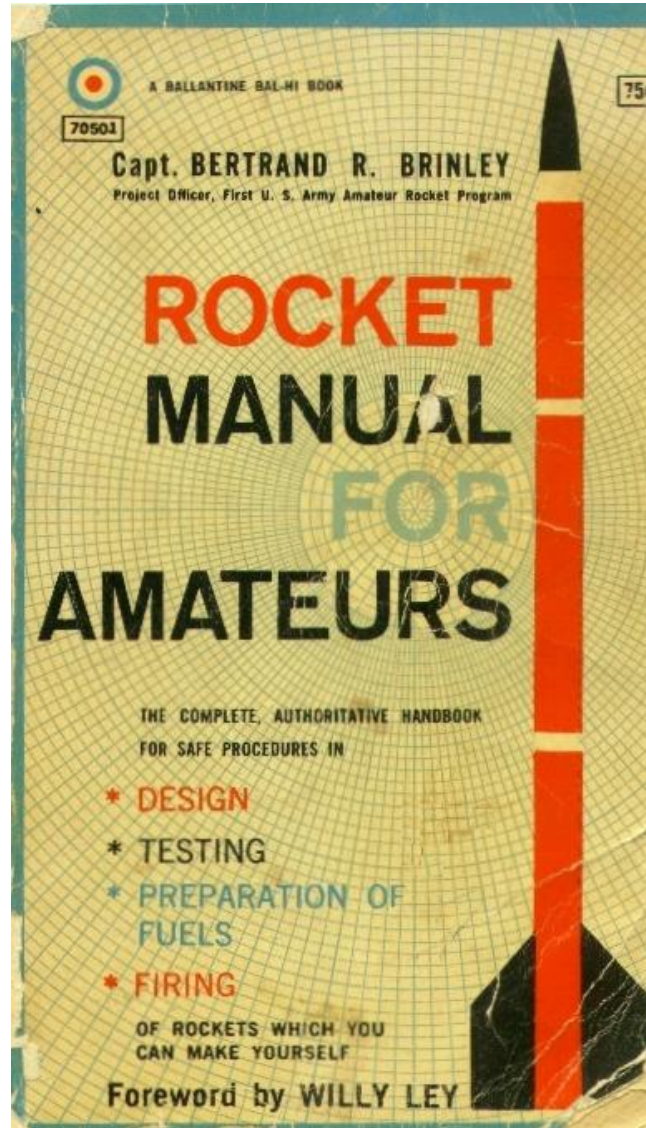
Recommended Engines (All four must be the same.)
A8-3, B6-4, C6-5

My early experience with AER

- Inspired to learn more about rockets and rocket motors, in particular.
- Very few books available on AER.

Amateur “bible” of the 1960’s

by Capt. B.R.Brinley, published 1960-1968



- Brinley's rockets were powered by zinc + sulfur propellant
- Brinley's book briefly described the **“Sugar Propellant”**.
- This led to the development of my first amateur rocket.

First amateur rocket launch, Feb.26, 1972



An old relic...

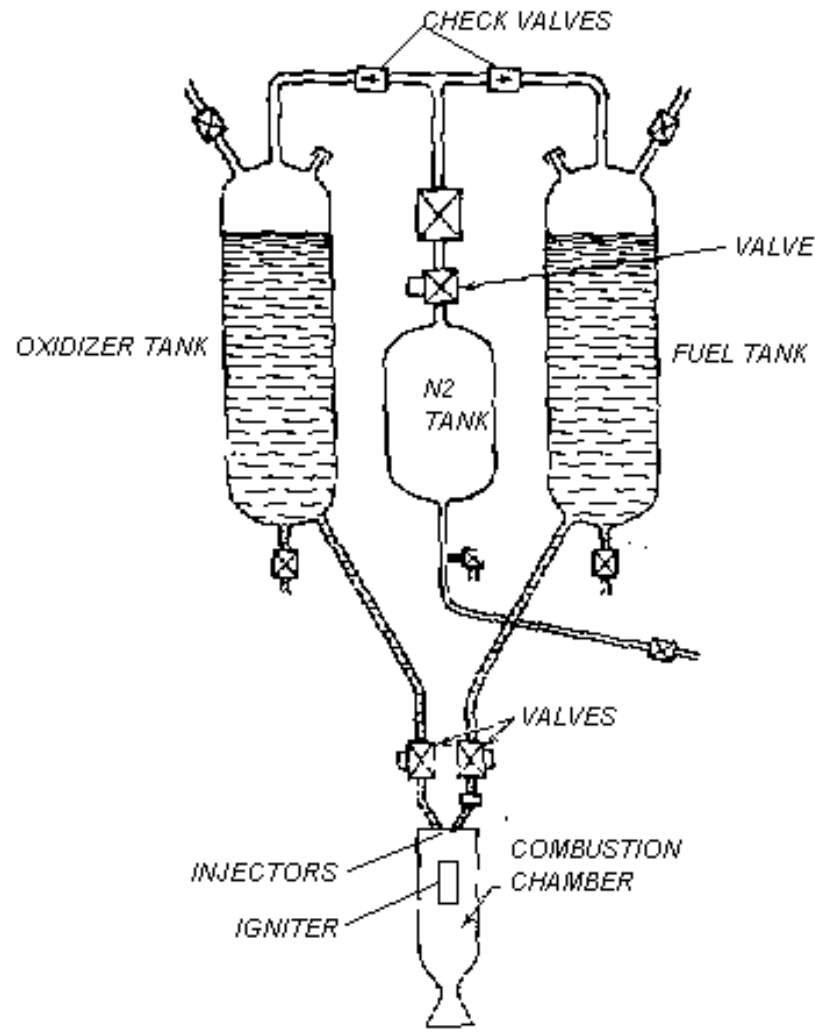


Types of Rocket Motors

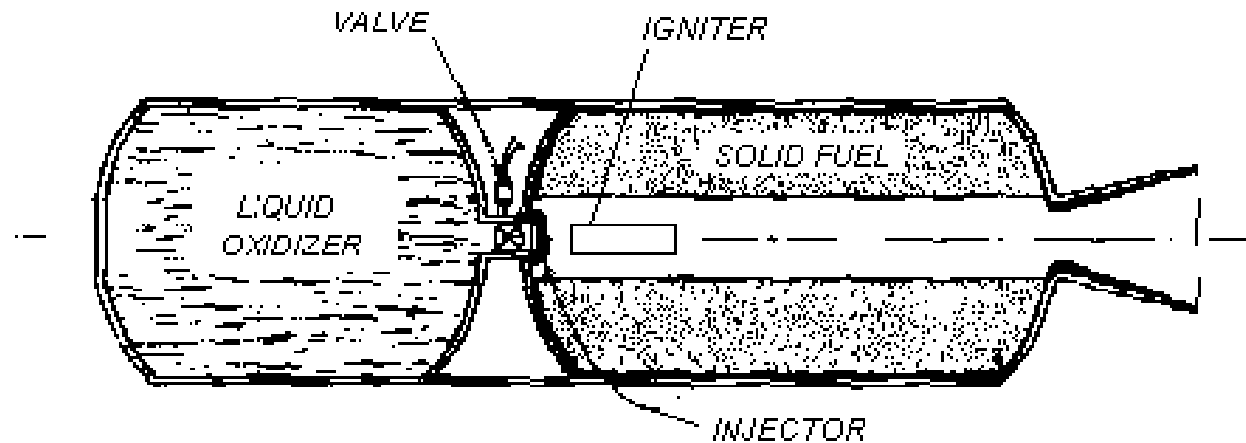
all are applicable to amateur rocketry

- Liquid propellant
- Hybrid
- Solid propellant

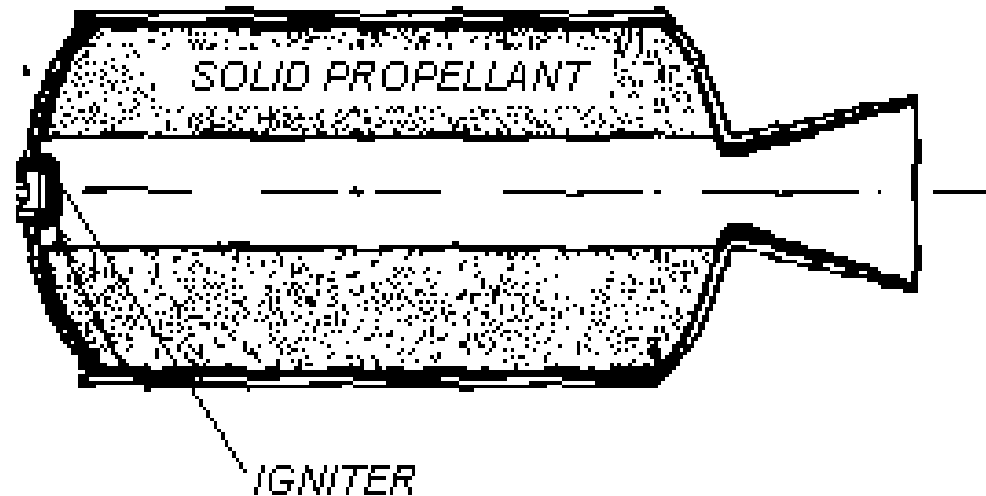
Liquid rocket motor



Hybrid rocket motor



Solid rocket motor



Solid Rocket Motors (SRM)



A-100M (“G” class motor)

Can be field reloaded in 10 minutes





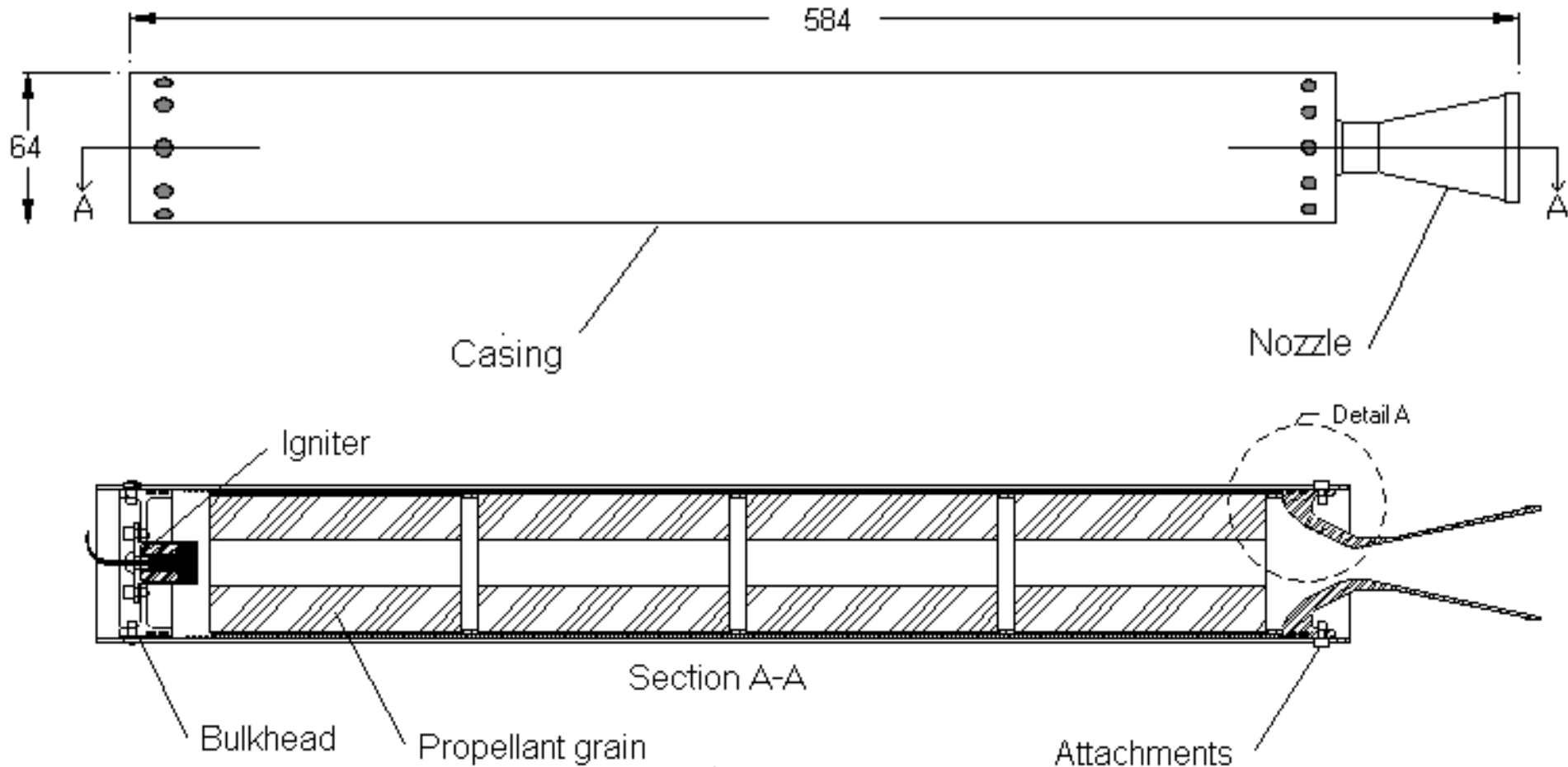
example: Liberty

“L” Class 3300 N-sec
3.2 kg. RNX propellant

“Rod & Tube” grain



Kappa engineering drawings

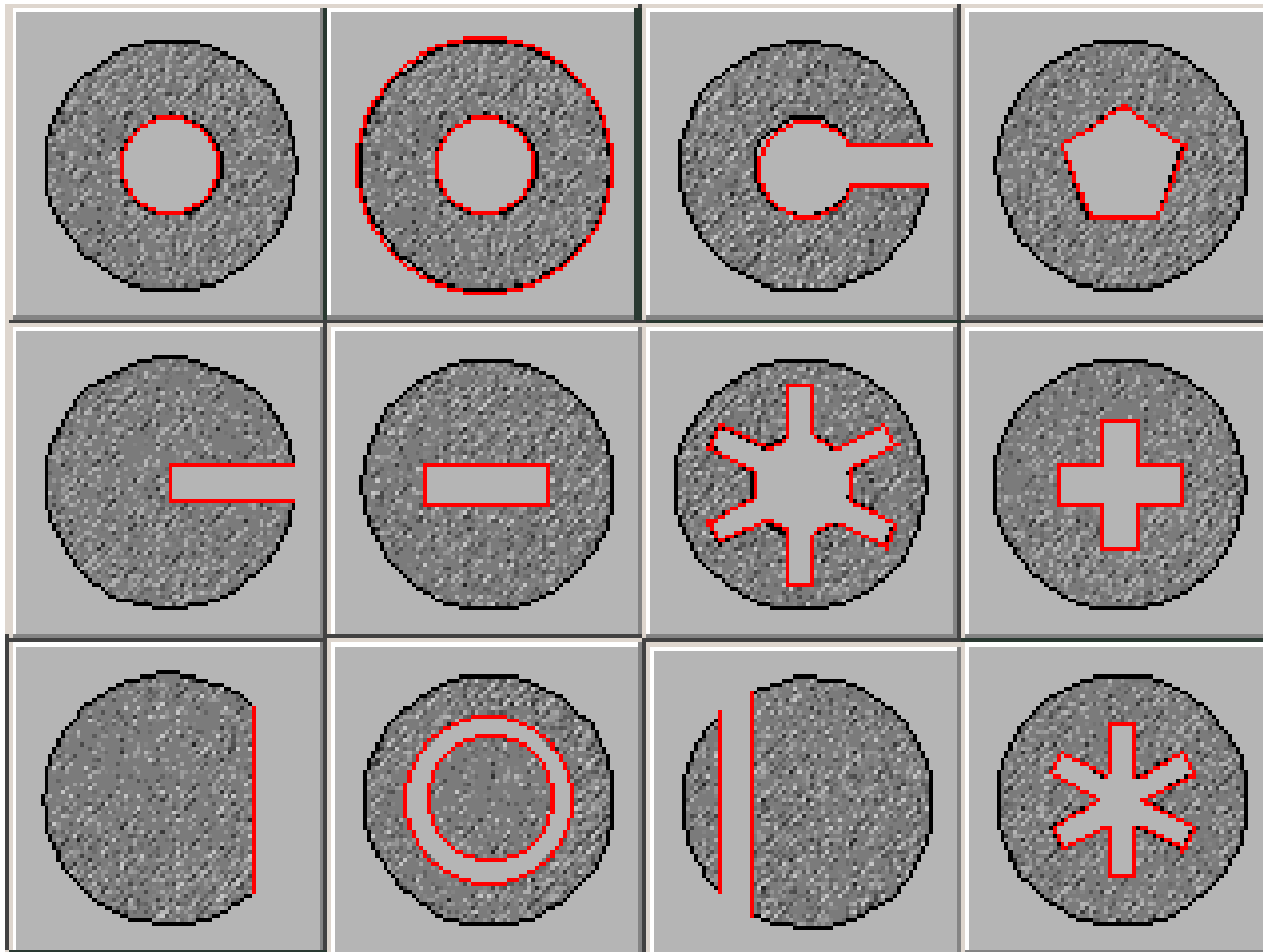


Propellant grain

Kappa-DX motor

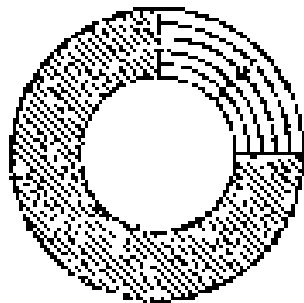


Grain configurations

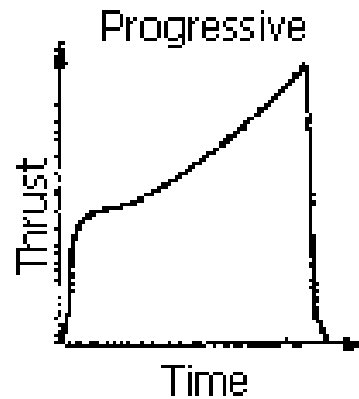


Red is
initial
burning
surface

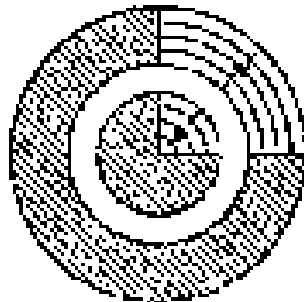
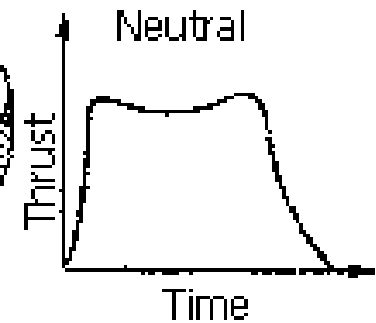
Grain configurations



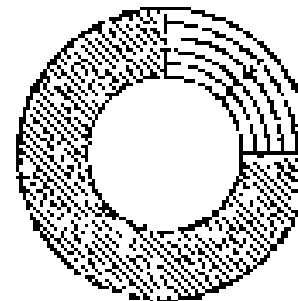
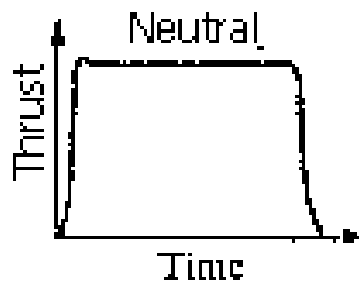
Tubular



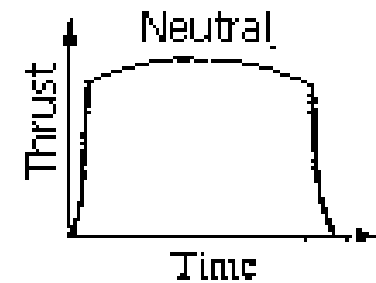
Star



Rod and Tube

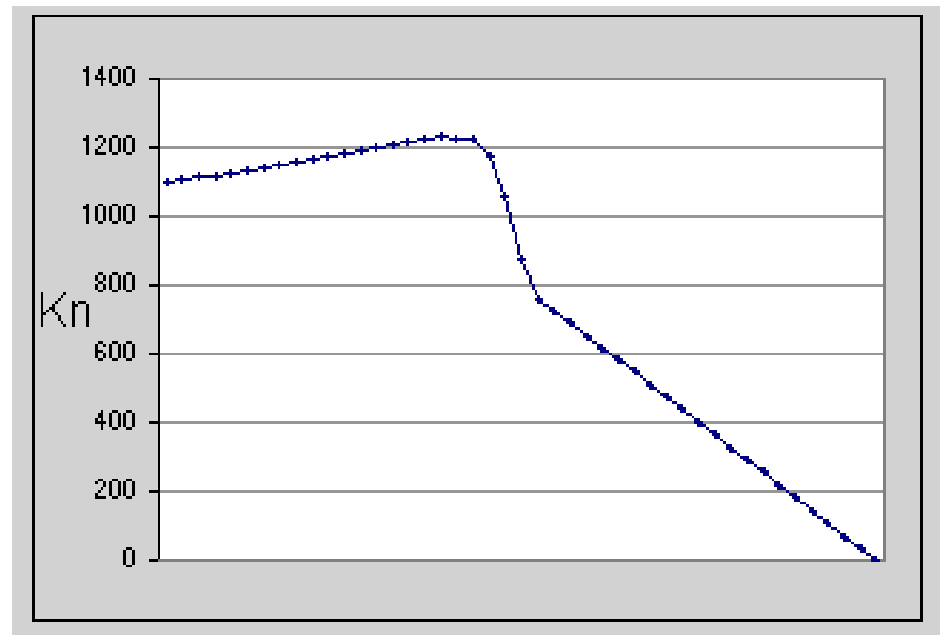


BATES
(multi-segment)



Grain configurations

Pseudo-finocyl



Grain configurations

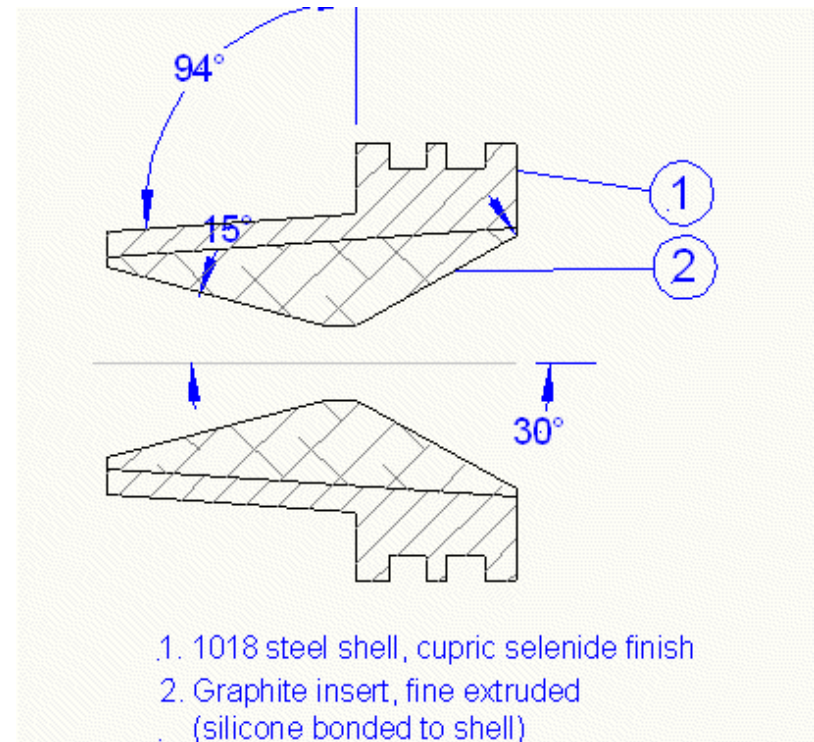
Rod & Tube



The Rocket Nozzle



Nozzles with graphite inserts



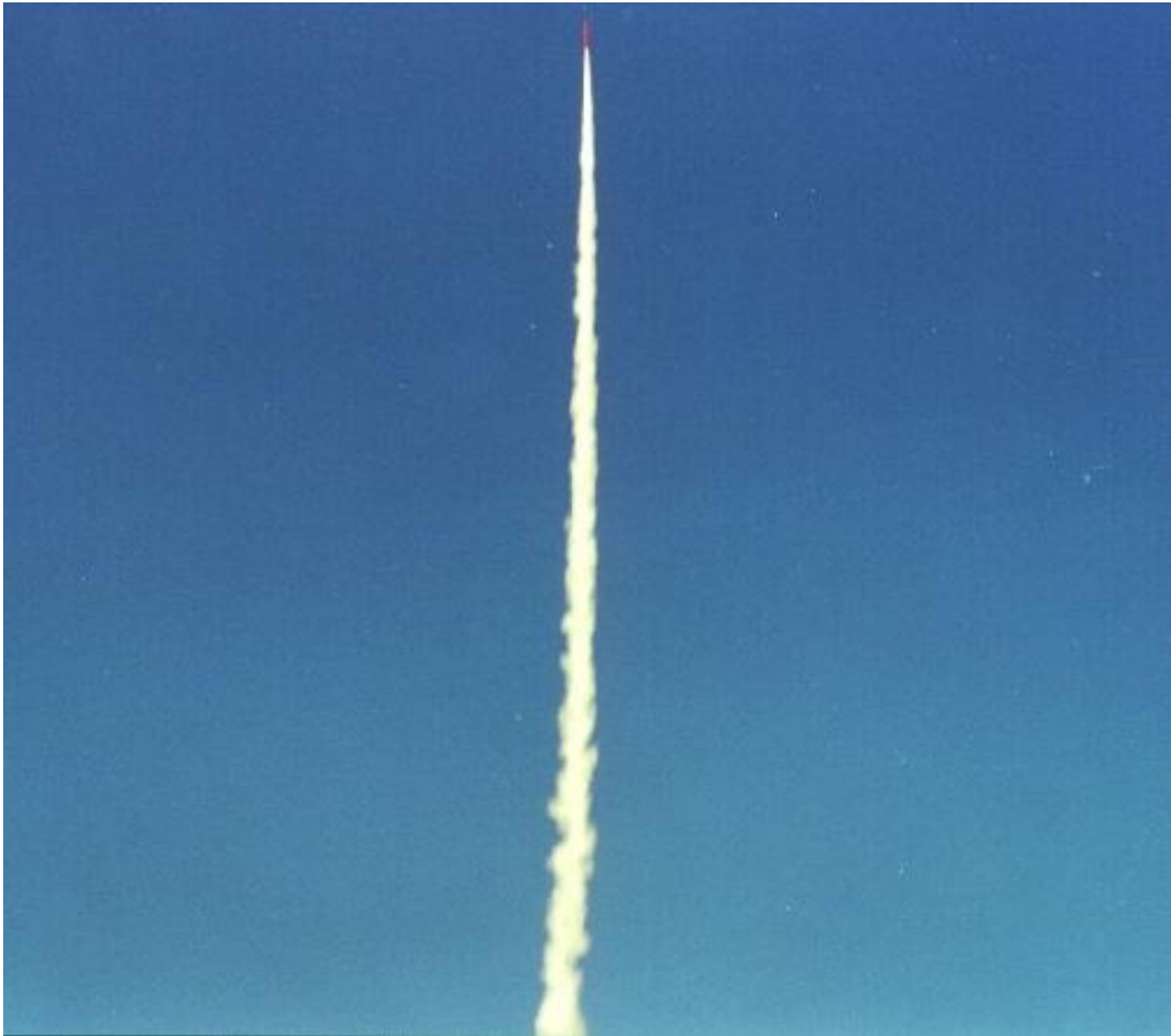
Rocket Flights

➤ Cirrus

➤ Frostfire



Cirrus One with “K” class KNDX propellant



VIDEO-CLIP

Cirrus One (2001)



VIDEO-CLIP

Frostfire 3 (2005)



VIDEO-CLIP

Frostfire 3 SLOW-MOTION



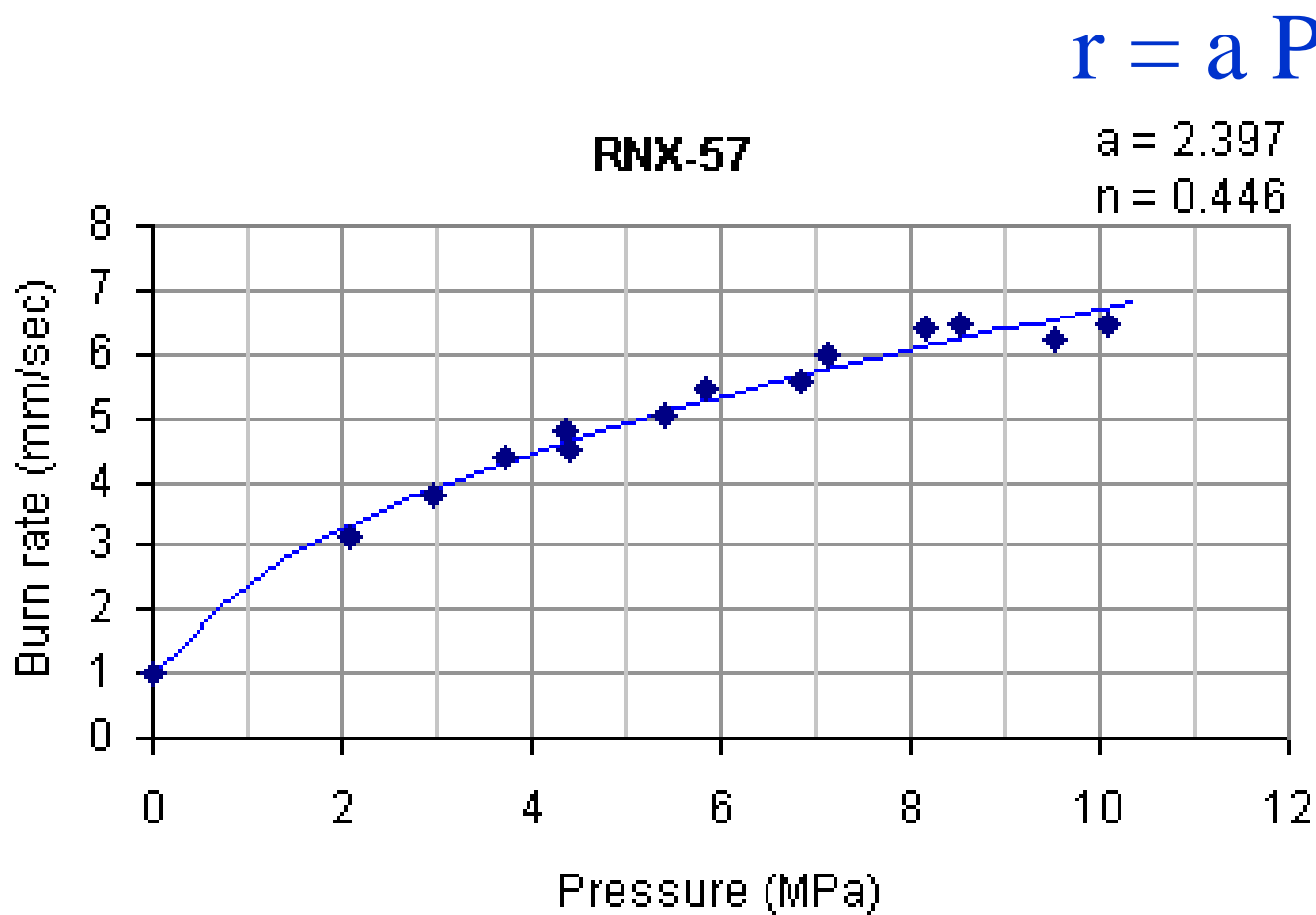
How is propellant burn rate
determined?

Burn rate measurement

Crawford Strand Burner



Burn rate measurement



Amateur Experimental Propellants:

“Classic” formulations:

1. Zinc/Sulfur (Micrograin)
2. Potassium Nitrate/Sucrose (“Sugar Propellant”)
3. Blackpowder (KN/Charcoal/Sulfur)

Amateur Experimental Propellants

Formulations that I have experience with

- **Potassium Nitrate/Sucrose (KNSU)**
- **Potassium Nitrate/Sorbitol (KNSB)**
- **Potassium Nitrate/Dextrose (KNDX)**
- **Potassium Nitrate/Fructose (KNFR)**
- **Potassium Nitrate/Epoxy/RIO (RNX)**
- **Ammonium Nitrate/Aluminum/Neoprene (ANAL)**

Sugar Propellants

Oxidizer: Potassium Nitrate , KNO_3

Fuel: Sucrose $\text{C}_{12}\text{H}_{22}\text{O}_{11}$

or Dextrose (glucose) $\text{C}_6\text{H}_{12}\text{O}_6$

or Fructose $\text{C}_6\text{H}_{12}\text{O}_6$

or Sorbitol $\text{C}_6\text{H}_{14}\text{O}_6$

or sugar alloys

RNX Epoxy Propellants

Oxidizer: Potassium Nitrate, KNO_3

Fuel: Epoxy: thermosetting resin

Epoxy consists of liquid Resin and liquid Hardener

Burn rate modifier: Red Iron Oxide: Fe_2O_3

Typical ratios, KN/Epoxy/RIO 70/22/8

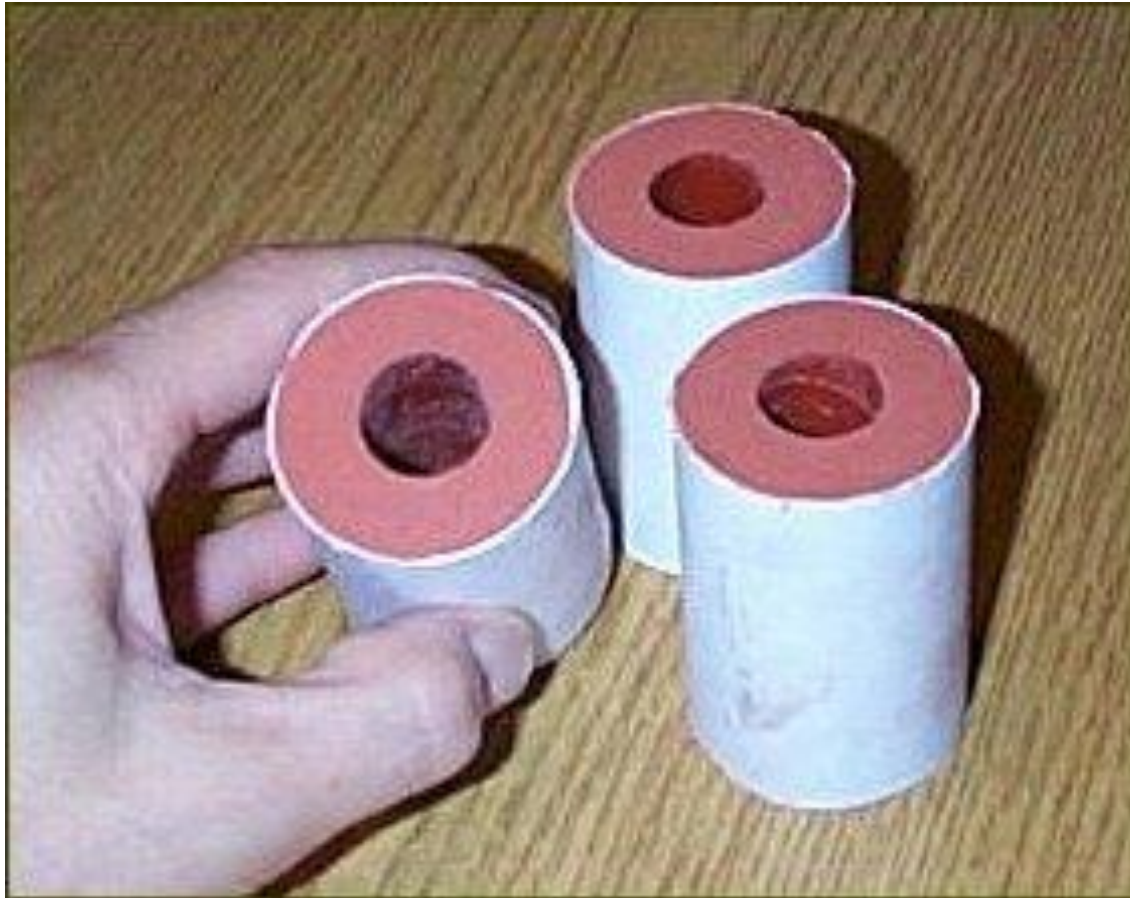
RNX Propellant Manufacture

- RNX epoxy propellants are cold cast
- KN and RIO are well blended first using tumbler
- Epoxy (liquid) added and mixed to a “putty” texture
- Packed into mould
- Cured at room temperature for 24 hours
- Can be safely cut, drilled or machined

RNX propellant mixing



RNX propellant finished product



Safety of manufacture

- Due to **heat casting** of **Sugar Propellant**, safety concern needs to be addressed.
- **“Overheating experiments”** conducted to address concern. Performed on KNSU, KNDX & KNSB.

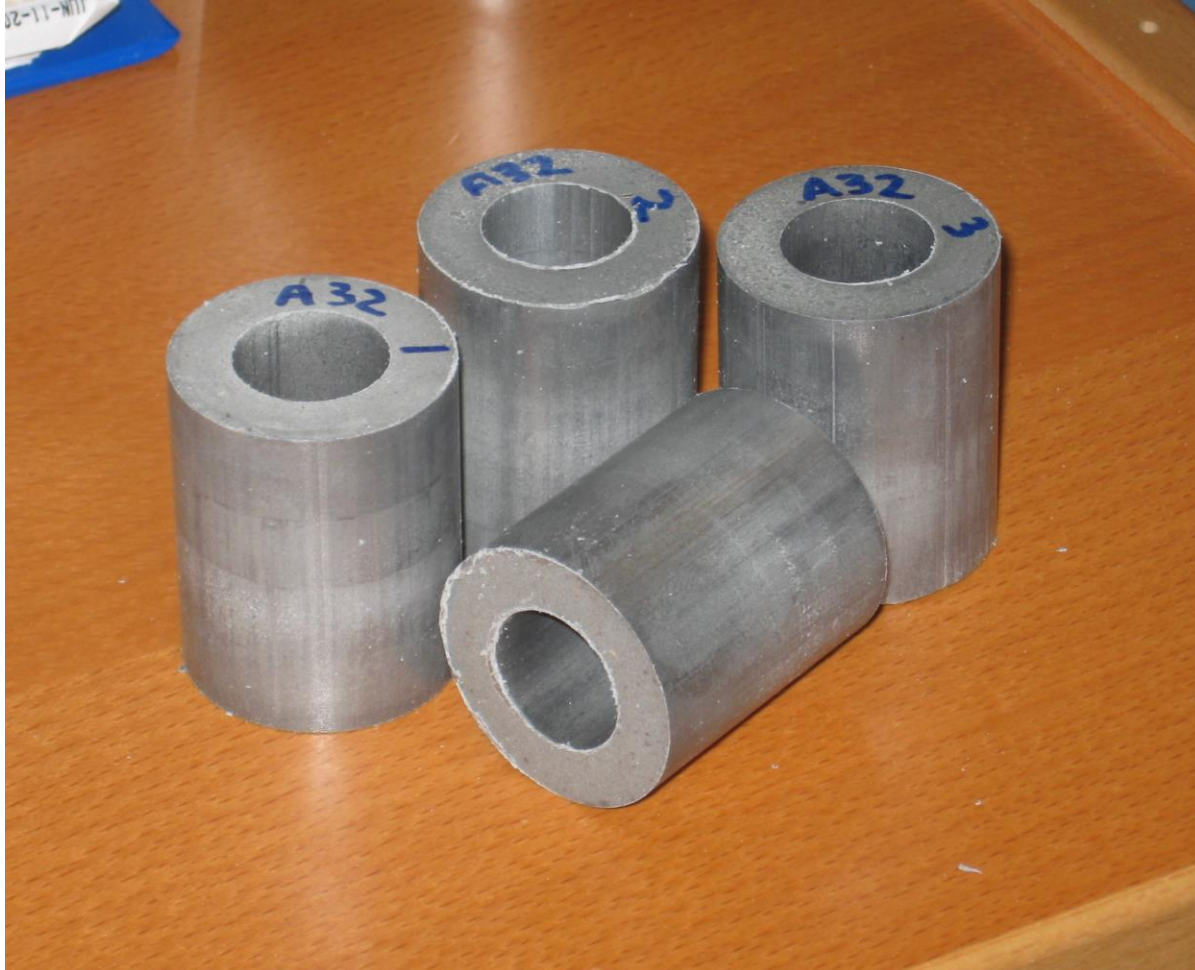
Safety of manufacture

- Thermal energy goes into decomposing the sugar, temperature remains approximately constant at 200° - 220° C. until decomposition is complete.
- Auto-ignition > 300°C.
- Casting: 130° - 185° C
- Large “safety margin” between casting temperature and auto-ignition temperature
- Obvious signs of overheating: discolouration, vapour, smell, etc.

Safety

- RNX propellant is **cold cast**.
- Uncured propellant is difficult to ignite. If ignited, burns weakly.
- Cured propellant is hard to ignite, is not friction or impact sensitive.
- No significant safety issues in manufacturing RNX propellant.

AN-Al Propellants



Rocket Motors

❖ Design and Testing

Internal Ballistics & Performance

- **SRM.xls** developed to simplify the safe design of amateur rocket motors.

Rocket Motor Structural Design

- Very important, as motor operates under high pressure
- Inadequate design can lead to **CATO**
- Casing *design strength & burst strength*
- Nozzle & bulkhead retention

Rocket Motor Structural Design

Closure attachment

- snap rings
- threaded closures
- radial pattern of machine screws
- bonded





Snap ring closure
retention



Rocket Motor Testing

- Verify design
- Provide actual performance numbers
- Reveal flaws in design or construction
- Provide reliability data

STS-5000 static test stand



Setup for Kappa KDX-002, July 22, 2000



VIDEO-CLIP

Kappa_KDX002_static.mpg



Load cells for force (thrust) measurement

Hydraulic

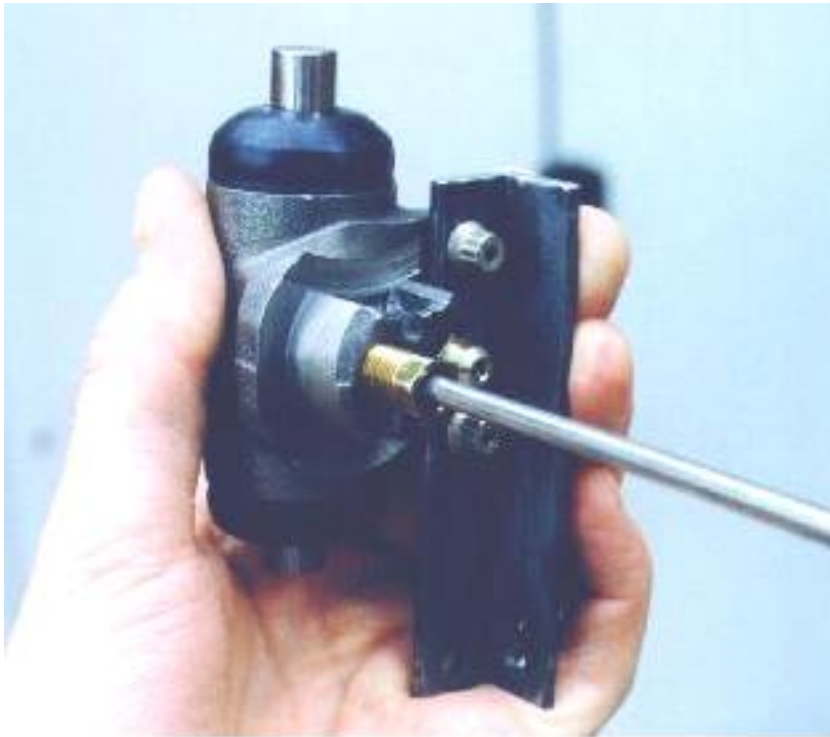
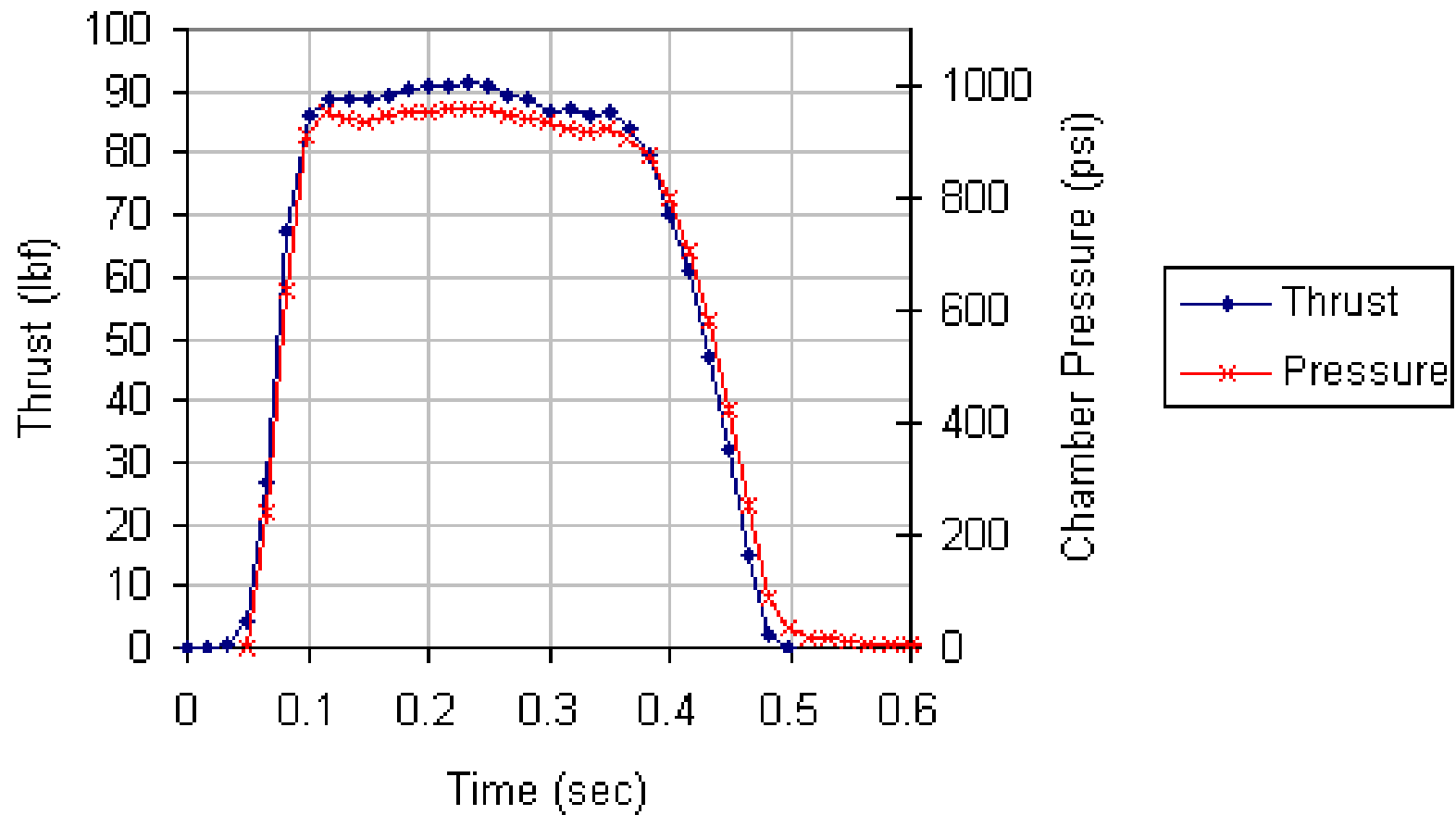


Figure 1 -- Hydraulic Load Cell

Electronic



Test results for A-100M motor, KNDX grain



From static testing, one can determine:

- Thrust as a function of time
- Chamber pressure as a function of time
- Total impulse
- Specific impulse
- Characteristic Velocity (c^*)
- Thrust Coefficient (C_f)
- Verify structural capability of motor
- Verify thermal capability of motor (e.g nozzle erosion, casing damage, insulation effectiveness)
- Anomalies in behaviour

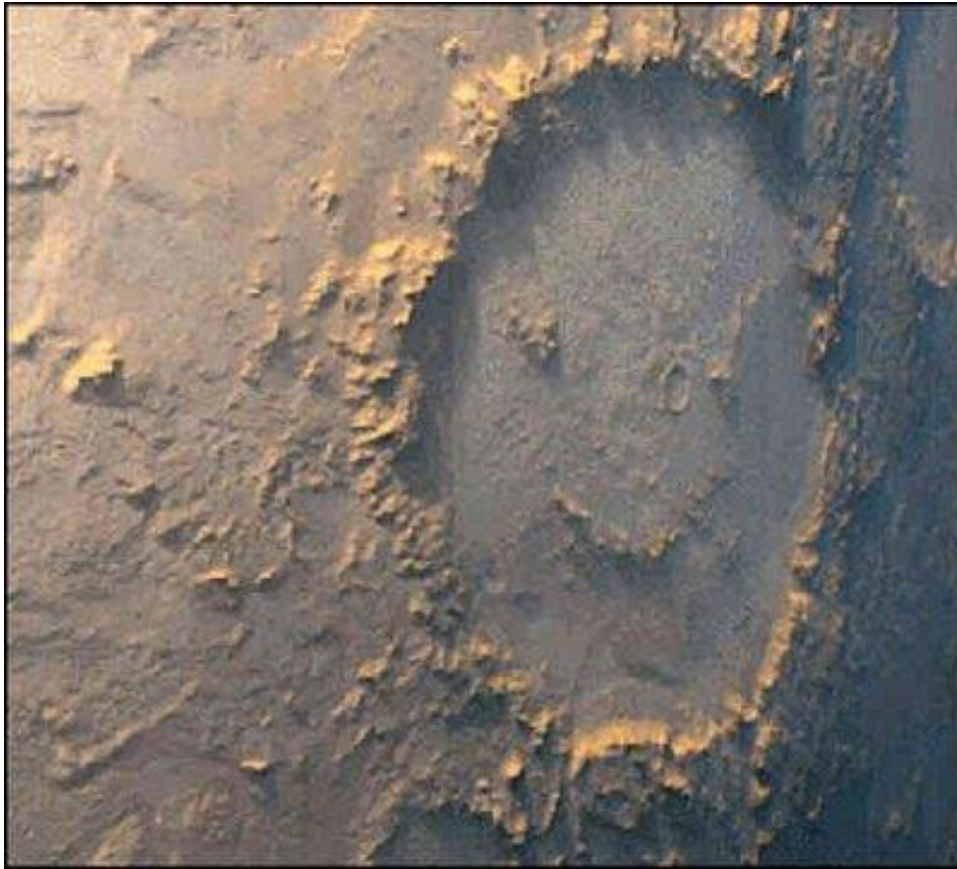
Concluding remarks

- What can be gained by involvement in rocketry?

- Rocketry is **multi-disciplinary**
- Chemistry, math, thermodynamics, aerodynamics, strength of materials, electronics, failure analysis, etc., etc.
- Hands-on work: machining, metal working, model building, composite techniques, experimental method, organizational skills
- Team-building amongst fellow enthusiasts.

Rocketry promotes an
interest in science &
engineering.

- And rocketry experience looks good on a resume.



❖ End